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EVALUATION OF FPC-1® FUEL PERFORMANCE CATALYST

AT

CHERAN STATE ROAD TRANSPORT CORPORATION

Report Prepared by

UHI CORPORATION PROVO, UTAH,

June 30, 1994

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INTRODUCTION

FPC-1* is a combustion catalyst which, when added to liquid hydrocarbon fuels, improves the combustion reaction resulting in increased engine efficiency and reduced fuel consumption. The products of incomplete combustion are also positively affected.

Field and laboratory tests alike indicate a potential to reduce fuel consumption in diesel fleets in the range of 5% to 10%. Smoke and carbon monoxide emissions are typically reduced 15 to 30%. This report summarizes the results of controlled back-to-back field tests conducted by UHI Corporation and Cheran State Road Transportation Corporation (CSRTC) engineers, with and without FPC-1* added to the high speed diesel fuel. The fuel consumption determination procedure applied was the <u>Carbon Balance Exhaust Emission Test</u> at a given engine load and speed. This same method also measures the exhaust concentrations of carbon monoxide and unburned hydrocarbons. Smoke testing was also conducted using the Bacharach Smokemeter method.

ENGINES TESTED

5 x Hinos

TEST INSTRUMENTS:

The equipment and instruments involved in the carbon balance test program were:

Sun Electric SGA-9000 non-dispersive, infrared analyzer (NDIR) for measuring the exhaust gas constituents, HC (unburned hydrocarbons as hexane gas), CO, CO₂, and O₂.

Scott Specialty BAR 90 calibration gases for SGA-9000 internal calibration of the SGA-9000.

A Fluke Model 51 type "k" thermometer and wet/dry probe for measuring exhaust, fuel, and ambient temperature.

A Dwyer magnehelic and pitot tube for exhaust pressure differential measurement and exhaust air flow determination (CFM).

Monarch Phototachometer and magnetic tape to determine and control engine speed (rpm).

A Bacharach True-Spot smokespot meter to determine the density of exhaust smoke from diesel engines.

A hydrometer and flask for fuel specific gravity (density) measurement.

A Hewlett Packard Model 42S programmable calculator for the calculation of the engine performance factors.

A Snap On throttle control for setting and holding engine speed at a fixed rpm.

TEST PROCEDURE

Carbon Balance

The carbon balance technique for determining changes in fuel consumption has been recognized by the US Environment Protection Agency (EPA) since 1973 and is central to the EPA-Federal Test Procedures (FTP) and Highway Fuel Economy Test (HFET). The method relies upon the measurement of vehicle exhaust emissions to determine fuel consumption rather than direct measurement (volumetric or gravimetric) of fuel consumption.

The application of the carbon balance test method utilized in this study involves the measurement of exhaust gases of a stationary vehicle under steady-state engine conditions. The method produces a value of engine fuel consumption with FPC-1* relative to a baseline value established with the same vehicle.

Engine speed and load are duplicated from test to test, and measurements of carbon containing exhaust gases (CO_2 , CO, HC), oxygen (O_2), exhaust and ambient temperature, and exhaust and ambient pressure are made. A minimum of five readings are taken for each of the above parameters after engine stabilization has taken place (rpm, and exhaust, oil, and water temperature have stabilized). The technical approach to the carbon balance method is detailed in the Appendices.

Fuel specific gravity or density is measured enabling corrections to be made to the final engine performance factors based upon the energy content of the fuel reaching the injectors.

Smoke density was determined by drawing a fixed quantity of exhaust gases through a filter medium. The particulate's were collected onto the filter surface and the density determined by comparing the discoloration of the filter paper to a color calibrated scale.

Five buses made up the final test fleet. Table 1 below summarizes the percent change in fuel consumption.

Table 1: Summary of Carbon Balance Fuel Consumption Changes

Bus #	Engine	Base RPM	FPC-1 RPM	% Change Fuel Consumption
*1498	Hino	2031	2025	- 0.78
1767	Hino	2075	2070	- 6.01
1584	Hino	1988	1980	- 4.86
1850	Hino	2055	2050	- 13.15
1443	Hino	2040	2050	- 4.94
1		2310		

^{*} Anomaly (see Discussion No. 3)

DISCUSSION

1. Fuel Density

High speed diesel was taken from the fuel tank on each bus to determine the fuel density (fuel specific gravity) for the baseline and treated fuel test segments. The fuel specific gravity for the treated test segment was generally higher than the baseline fuel specific gravity. The correction factor for each bus is shown on the computer printouts which also show the calculation of the baseline and FPC-1 treated fuel performance factors (or mass flow rates). The correction factor adjusts the energy content of the treated fuel to that of the baseline fuel.

2. The Effect of FPC-1 upon Smoke Density

Smoke density was determined using the Bacharach smoke spot method. The Bacharach True-Spot Smokemeter measures smoke density by drawing a specific volume of exhaust gas through a fine paper filter medium (5 micron) while the engine is operating at a fixed rpm and under steady-state engine conditions. The smoke particles are trapped on the surface of the filter paper as the exhaust gases are drawn through it forming a darkened area called a "smoke spot". The filter paper is then removed from the smoke tester and the smoke spot visually compared to a precoded smoke scale. A smoke number is then assigned to the smoke spot according to the darkness of the spot. The smoke number scale ranges from 0 to 9. Higher smoke numbers correspond to darker smoke spots, which correspond to a greater smoke density in the exhaust. The baseline and treated fuel smoke spot numbers are tabled below.

 Table 2:

 Comparison of Smoke Spot Numbers (SS#)

Bus #	Base SS#	Treated SS#	% Change
*1498	2.2	5.5	+150
1767	7.5	3.0	- 60
1584	5.5	6.0	+ 8
1850	8.2	6.0	- 27
1447	7.7	3.0	- 61

^{*} Anomalies (see Discussion Number 3.)

A reduction in smoke is prime evidence of improved combustion (Germane, SAE Technical Paper # 831204). Further, reduced exhaust smoking has been shown to be one of first evidences that engine carbon residue and soot blowby into the motor oil are also being reduced (ibid). The reductions in exhaust smoke are logical extensions of improved combustion created by FPC-1.

3. Anomalies

Bus number 1498 had the nozzles replaced between the baseline and treated segments of the test. The data indicate that the nozzle change had a detrimental impact upon engine performance since there was little change in fuel consumption and a large increase in smoke density. In any case, this type of mechanical alteration renders the data from Bus 1498 invalid.

CONCLUSIONS

- 1) With the anomalies removed from the sample, the fuel consumption change determined by the carbon balance method ranged from 4.86 to 13.15%. The fleet averaged a 7.24% reduction in fuel consumed after FPC-1 fuel treatment.
- 2) Smoke density, with anomalies removed, was reduced approximately 35%.

APPENDICES

CARBON BALANCE METHOD TECHNICAL APPROACH:

All test instruments were calibrated and zeroed prior to both baseline and treated fuel data collection. The SGA-9000 NDIR exhaust gas analyzer was internally calibrated using Scott Calibration Gases (BAR 90 Gases), and a leak test on the sampling hose and connections was performed. The same procedure was repeated after each test segment to determine any instrument drift.

Each vehicle's engine was brought up to operating temperature at a set rpm and allowed to stabilize as indicated by the engine water and exhaust temperature, and exhaust pressure. No exhaust gas measurements were made until each engine had stabilized at the rpm selected for the test. Engine rpm was set using the dash mounted tachometer (with the exception of shovel's #1 and #4) and checked periodically to prevent any change in engine speed during the data collection period. #2 diesel was used exclusively throughout the evaluation. Fuel specific gravity (density) and temperature were also taken.

The baseline fuel consumption test consisted of a minimum of five sets of measurements of CO_2 , CO, HC, O_2 , and exhaust temperature and pressure made at 90 second intervals. Each engine was tested in the same manner. Engine rpm were also recorded at approximately 90 second intervals.

After the baseline test the fuel storage tanks were treated with FPC-1* at the recommended level of 1 oz. of catalyst to 40 gallons of fuel (1:5000 volume ratio). Each succeeding fuel shipment was also treated with FPC-1*. The equipment was operated on treated fuel until the final test was run.

During the two test segments, an internal self-calibration of the exhaust analyzer was performed after every two sets of measurements to correct instrument drift, if any.

From the exhaust gas concentrations of CO_2 , CO, HC, and O_2 measured during the test, the average molecular weight of these gases, and the temperature and volumetric flow rate of the exhaust stream, the mass flow rate of the fuel to the engine (rate of fuel consumption) may be expressed as a engine "performance factor" which relates the fuel consumption of the treated fuel to the baseline. The calculations are based on the assumption that engine operating conditions are essentially the same throughout the test. Engines with known mechanical problems or having undergone repairs affecting fuel consumption are removed from the sample.

A sample calculation is found in Figure 2.

Details of Analysis

Figure 1

CARBON MASS BALANCE FORMULAE

ASSUMPTIONS: C_8H_{18} and SG = 0.78

Time is constant Load is constant

DATA:

Mwt = Molecular Weight

pf1 = Calculated Performance Factor (Baseline) pf2 = Calculated Performance Factor (Treated)

PF1 = Performance Factor (adjusted for Baseline exhaust mass)
PF2 = Performance Factor (adjusted for Treated exhaust mass)

CFM = Volumetric Flow Rate of the Exhaust

SG = Specific Gravity of the Fuel

VF = Volume Fraction

d = Exhaust stack diameter in inches Pv = Velocity pressure in inches of H_20

 P_B = Barometric pressure in inches of mercury

Te = Exhaust temperature ^oF

VFHC = "reading" \div 1,000,000

VFCO = "reading" ÷ 100 VFCO₂ = "reading" ÷ 100 VFO₂ = "reading" ÷ 100

EQUATIONS:

$$Mwt = (VFHC)(86) + (VFCO)(28) + (VFCO2)(44) + (VFO2)(32) + [(1-VFHC-VFCO-VFCO2-VFO2)(28)]$$

pf1 or pf2 =
$$\frac{2952.3 \text{ x Mwt}}{86(\text{VFHC}) + 13.89(\text{VFCO}) + 13.89(\text{VFCO}_2)}$$

CFM =
$$(d/2)^2 \pi / 144$$
 • 1096.2 $\frac{Pv}{1.325 (P_B/ET + 460)}$

$$PF1 \text{ or } PF2 = \frac{\text{pf x (Te+460)}}{CFM}$$

FUEL ECONOMY:
PERCENT INCREASE (OR DECREASE)

PF2 - PF1 _____ x 100 _____ x 100

SAMPLE CALCULATION FOR THE CARBON MASS BALANCE

BASELINE:

Equation 1 (Volume Fractions)

VFHC = 13.20/1,000,000 = 0.0000132

VFCO = 0.017/100

= 0.00017

 $VFCO_2 = 1.937/100$ = 0.01937

 $VFO_2 = 17.10/100$ = 0.171

Equation 2 (Molecular Weight)

 $\begin{aligned} \text{Mwt1} &= (0.0000132)(86) + (0.00017)(28) + (0.01937)(44) + (0.171)(32) \\ &+ [(1-0.0000132 - 0.00017 - 0.01937 - 0.171)(28)] \end{aligned}$

Mwt1 = 28.995

Equation 3 (Calculated Performance Factor)

$$pf1 = \underline{2952.3 \times 28.995} \\ 86(0.0000132) + 13.89(0.00017) + 13.89(0.01937)$$

pf1 = 314,083

Equation 4 (CFM Calculations)

CFM =
$$\frac{(d/2)^2 \pi}{144}$$
 • 1096.2 $\frac{Pv}{1.325 \{P_B/(ET + 460)\}}$

d = Exhaust stack diameter in inches

Pv = Velocity pressure in inches of H_20

P_B =Barometric pressure in inches of mercury

Te =Exhaust temperature ^oF

CFM =
$$\frac{(10/2)^2\pi}{144}$$
 • 1096.2 $\frac{.80}{1.325\{30.00/(313.100 + 460)\}}$

$$CFM = 2358.37$$

Equation 5 (Corrected Performance Factor)

$$PF1 = 314,083 (313.1 \text{ deg } F + 460)$$

$$2358.37 \text{ CFM}$$

$$PF1 = 102,960$$

TREATED:

Equation 1 (Volume Fractions)

$$VFCO_2 = 1.826/100$$

= 0.01826

$$VFO_2 = 17.17/100$$

= 0.1717

Equation 2 (Molecular Weight)

$$Mwt2 = (0.0000146)(86) + (0.00013)(28) + (0.01826)(44) + (0.1717)(32) + [(1-0.0000146-0.00013-0.01826-0.1717)(28)]$$

Mwt2 = 28.980

Equation 3 (Calculated Performance Factor)

$$pf2 = \underbrace{2952.3 \times 28.980}_{86(0.0000146) + 13.89(0.00013) + 13.89(0.01826)}$$

pf2 = 333,308

Equation 4 (CFM Calculations)

CFM =
$$\frac{(d/2)^2\pi}{144}$$
 • 1096.2 $\frac{Pv}{1.325 \{P_B/(Te + 460)\}}$

d = Exhaust stack diameter in inches

Pv = Velocity pressure in inches of H_20

 P_B =Barometric pressure in inches of mercury

Te = Exhaust temperature ^oF

CFM =
$$\frac{(10/2)^2\pi}{144}$$
 • 1096.2 $\frac{.775}{1.325\{29.86/(309.02 + 460)\}}$

CFM = 2320.51

Equation 5 (Corrected Performance Factor)

= 110.459

Fuel Specific Gravity Correction Factor

Baseline Fuel Specific Gravity - Treated Fuel Specific Gravity/Baseline Fuel Specific Gravity +1

$$.840 - .837 / .840 + 1 = 1.0036$$

$$PF2 = 110,459 \times 1.0036$$

$$PF2 = 110,857$$

Equation 6 (Percent Change in Engine Performance Factor:)

% Change PF =
$$[(110,857 - 102,960)/102,960](100)$$

= $+7.67$

Note: A positive change in PF equates to a reduction in fuel consumption.

RAW DATA WORK SHEETS

Company: Cheran Test Portion: Baseline:			e: 05 126 194. ck Diameter: Inches
restroition. Dascine.			Diminiciainches
		233100	
Engine Make/Model: _	Hino.	Miles/Hours:	_I.D.#: <u>185</u>
Type of Equipment:		1450	ANC. 87.2.
Fuel Specfic Gravity:	. 833	_ (0,7	6 @:(°F)
Barometric Pressure: _	gul	inches of Mercury	Start Time: 0105

RPM	Exhaust Temp E	P Inches of H ₂ O	% CO	HC ppm	% CO₂	% O ₂	NO.	
2050	206	.36	0.04	. 28	1.31	18.1		(4)
	206		0.01	- 26	1.31	18.1.		
	206.		0.04	27	1.34	18.2		
	206		0-04.	26	1.34	18.2.		
	206		0.04	28.	1.34	182		
	206		0.04	28	1.34	18.2		
	206		0.04.	28	1.34	18.2		
	206-		0.00	27	1.38	18.2		
	206		0.04	26	1.3.15	18.2		
2050	206.	. 36	0-00	27	1:33	18.2		

Names of Customer Personnel Participating in Test:

Fh/54 0120.

34 · 04 27 1.338 18.68 (,9336)

Company: Cheran, Loca Test Portion: Baseline:	ation: Test Date Treated: Exhaust Stack	: 05 25 Q + k Diameter:Inches	-los/26/94.
Engine Make/Model: Ley long Type of Equipment:	1370 Miles/Hours: 91,510	PI.D.#: 1443 43 - And	Temp 85.4.
Fuel Specfic Gravity:	0.70	<u>@: 89.2</u>	,
Barometric Pressure:	inches of Mercury	Start Time:	0230

RPM	Exhaust Temp %	P Inches of H ₂ O 4	% CO	HC ppm	% CO ₂	% O ₂	NO.
2050	226	· 40	0.07	24	1.68	17:5.	
1	220		0.09	23	1.69	17.6	
	220		0.05	27	1.70	175.	
	22 ₀		0.05	28	1.70-	175	
<u> </u>	220		0.05	27	1.70	17.5	
	220		0.00	28.	1.70	17.5	
e .	220		B . 0A	28	1.70	17.5	
	220		0.04	28	1.70	17.5.	
	220.		0 04.	25.	1.70-	17.4	
	220	· 40.	0.04	28	1.70	17.4	^

Names of Customer Personnel Participating in Test:

Finish.

224.0 -41 ,048 7728 1.697 12.49

_ inches of Mercury

Barometric Pressure:

Carbon Mass Balance Field Data Form		1584
Company: Cheran Location: CBE Test Date: OSI	26/94.	(6) (6)
Test Portion: Baseline: Treated: Exhaust Stack Diameter	er:Inches	
Engine Make/Model: Leyland 370 Miles/Hours: I.D.#:		
Type of Equipment:	- AIV	85.4
Fuel Specific Gravity: 0 833	D:(°F))

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
1989	226.	0.013	6).04	27	1.64.	17.5.	-
	227		0.02	27	1.62	17.5.	
	225	0.120	0.04	27	1.62	174.	
	228		0-04	27	1.62	17.4	•
	228		0 04	27	1.62	17.5.	
	228	,	0.04	27	1.62	17.5.	
	228	CIRC	0.04	26.	1.62	17.5	
	228		0.04	28	1.62	17.5	
	227		0.03	28	1.62	17.5	
	227	0-120	0.03	30	1.62	17.5	-

Smole.

Start Time: 0 1 40

Names of Customer Personnel Participating in Test:

27.4 1.622 17.48 227.2

History of Testing at Cheran, Andhra Pradesh, and Gujarat

On January 19, 1994, a team of technicians from UHI Corporation traveled to India at the request of the Petroleum Conservation Research Association (PCRA) to conduct field trials of FPC-1 in bus fleets operated by Gujarat and Andhra Pradesh State Road Transort Corporations. Mr. Natarajan had also arranged for a test at Coimbotore, for Cheran Road Transort Corporation. This was not a PCRA sponsored test, but something arranged through local sales people (Cheran is not a State Road Transport Corporation because it only covers a small portion of the State of Tamil Nadu, whereas the others service an entire State). The Andhra Pradesh test was not conducted due to communication problems, but the tests were conducted at Gujarat and Cheran.

The Cheran test was conducted on February 5, 1994. Twelve buses were baselined using carbon mass balance. All pertinent personnel, including the Chief Mechanical Officer and Fleet Manager, as well as dozens of mechanics and staff engineers.

The Gujarat test was conducted on February 11, 1994. Eight buses were baselined using carbon mass balance. Again, all managers attended the test.

The fuel was treated approximately one week after the baseline for both fleets. The test buses ran on treated fuel until May of 1994, when the UHI team returned to conduct the treated tests.

The Cheran treated was conducted May 26, 1994. Only five buses were available for the treated segment. The Gujarat test was conducted May 31, 1994. All eight buses were available for this test.

Fuel consumption reductions averaged 7.24% for Cheran, and 6.56% for Gujarat. Smoke reductions averaged 35% and 27.7%, respectively. FPC-1 had little effect upon carbon monoxide emissions, indicating the engines may not have achieved complete breakin before final testing.

During both trips to India by UHI, visits were made to Mr. Ghosh and Mr. Das of the PCRA. Mr. Agarwal of the Indian Railways was also visited. Both groups were given a full two hour presentation on FPC-1, as were the engineering staffs at Cheran, Gujarat, and Andhra Pradesh.

Also, during the second trip to India, Bob Platt and I conducted the baseline for Andhra Pradesh (approx. June 2nd). Bob has since completed that study and has forwarded a final report to UHI.

Figure 1

CARBON MASS BALANCE FORMULAE

ASSUMPTIONS: $C_{12}H_{26}$ and SG = 0.78

> Time is constant Load is constant

Mwt = Molecular Weight DATA:

> = Calculated Performance Factor (Baseline) pf1 = Calculated Performance Factor (Treated) pf2

PF1 = Performance Factor (adjusted for Baseline exhaust mass) PF2 = Performance Factor (adjusted for Treated exhaust mass)

CFM = Volumetric Flow Rate of the Exhaust

SG = Specific Gravity of the Fuel

VF = Volume Fraction

d = Exhaust stack diameter in inches Pv = Velocity pressure in inches of H₂0

 P_{B} = Barometric pressure in inches of mercury

= Exhaust temperature ^oF ET

VFHC = "reading" \div 1,000,000

 $\begin{array}{lll} \text{VFCO} & = \text{"reading"} \div 100 \\ \text{VFCO}_2 & = \text{"reading"} \div 100 \\ \text{VFO}_2 & = \text{"reading"} \div 100 \end{array}$

EQUATIONS:

$$Mwt = (VFHC)(86) + (VFCO)(28) + (VFCO2)(44) + (VFO2)(32) + [(1-VFHC-VFCO-VFCO2-VFO2)(28)]$$

pf1 or pf2 =
$$\frac{2952.3 \text{ x Mwt}}{86(\text{VFHC}) + 13.89(\text{VFCO}) + 13.89(\text{VFCO}_2)}$$

CFM =
$$(d/2)^2 \pi / 144$$
 • 1096.2 $\frac{Pv}{1.325 (P_B/ET + 460)}$

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Leyland 665	Mile/Hrs	107628			
Equipment Type:	Bus	ID #:	GJ1Z2750		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2			
					Time:	935

RPM	Exh Temp	Pv Inch	CO	HC	€02	O2	
2016	214.4	0.58	0	26	1.49	18.1	
2016	214.4	0.6	0	26	1.49	18.1	
2016	214.4	0.56	0	19.5	1.46	18.2	
2026	215	0.6	0	23	1.44	18.2	
2030	214	0.6	0	28.5	1.44	18.3	
2030	212.6	0.58	0	24	1.43	18.3	
2022.333	214.133	.587	.000	24.500	1.458	18.200	Mean
7.089898918	0.816496581	0.01632993	0	3.09838668	0.02639444	0.08944272	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
2.45E-05	0	0.01458333	0.182	28.9627543	417,630	3,618,369

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/3194
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Leyland 665	Mile/Hrs:	181190			
Equipment Type	Bus	ID #:	GJ1Z2750	•	Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	104			
SG Corr Factor:	1.01				Time:	1635

RPM	Exh Temp	Pv Inch	CO	HC	€02	O2	
2024	235	0.6	0	31	1.22	18.2	
2024	235	0.6	0	30	1.23	18.2	
2024	235.6	0.6	0	31	1.22	18.2	
2024	235.8	0.6	0	27.5	1.22	18.2	
2024	236	0.6	0	31	1.22	18.2	
2024	237	0.44	0	27.5	1.25	18.2	
2024	237	0.5	0	29	1.23	18.3	
2024	237.4	0.5	0	25	1.23	18.3	
2024	238	0.5	0	28	1.22	18.3	
2024.000	236.311	.549	.000	28.889	1,227	18.233	Mean
0	1.077548658	0.06333333	0	2.07330922	0.01	0.05	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.89E-05	0	0.01226667	0.182333333	28.9272756	493,781	4,495,091

4,540,042

**% Change PF=

25.47 %

Company Name:	Gujart	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	665 New	Mile/Hrs	52240			
Equipment Type:	Bus	ID #:	GJ1Z3125		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2		Time:	1255

RPM	Exh Temp	Pv Inch	CO	II(C	CO2	02	
2007	213	0.52	0	27.5	1.33	18.4	
2010	213	0.54	0	46	1.33	18.4	
2010	209.6	0.54	0	28.5	1.32	18.4	
2010	208	0.52	0	31	1.32	18.4	
2010	208	0.54	0	29	1.31	18.4	
2010	207.8	0.54	0	29.5	1.3	18.4	
							-
2009.500	209.900	.533	.000	31.917	1.318	18.400	Mean
1.224744871	2.487569095	0.01032796	0	6.99583209	0.01169045	0	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
3.19E-05	0	0.01318333	0.184	28.9487845	459,598	4,163,208

					Carlotte of the control of		
Company Name:	Gujart	Location:	Ambaji		Test Date:	5/31/94	
Test Portion:	Treated	Stack Diam:	2	Inches			
Engine Type:	665 New	Mile/Hrs:	130518				
Equipment Type	Bus	ID #:	GJ1Z3125		Baro:	28.17	
Fuel Sp. Gravity:	0.824	Temp:	100				
SG Corr Factor:	1.01				Time:	1235	

RPM	Exh Temp	Pv Inch	CO	II(e	CO2	02	
2015	233	0.52	0	24.5	1.23	18	
2015	233	0.54	0	26.5	1.2	18	
2015	229	0.54	0	25	1.2	18.1	
2015	238.5	0.54	0	27	1.2	18.2	
2015	236.5	0.54	0	26	1.19	18.2	
2015	238	0.56	0	25.5	1.18	18.3	
2015.000	234.667	.540	.000	25.750	1.200	18.133	Mean
0	3.656045222	0.01264911	0	0.93541435	0.0167332	0.12110601	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.58E-05	0	0.012	0.181333333	28.9188268	505,274	4,631,939

4,678,258

**% Change PF=

12.37 %

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Wiking BW	Mile/Hrs	47597			
Equipment Type:	Bus	ID #:	GJ1Z3029		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2		Time:	

RPM	Exit Temp	Pv Inch	C ()	HC	CO2	O2	
2011	213.4	0.6	0	27	1.42	18.2	
2011	213.6	0.56	0	30.5	1.41	18.2	
2016	221.2	0.56	0	29.5	1.4	18.2	
2016	217	0.58	0	37.5	1.38	18.2	
2016	217.2	0.58	0	32.5	1.39	18.3	
2016	217	0.54	0	28	1.37	18.3	
2014.333	216.567	.570	.000	30.833	1.395	18.233	Mean
2.581988897	2.86612398	0.02097618	0	3.7903386	0.01870829	0.05163978	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
3.08E-05	0	0.01395	0.182333333	28.9543217	435,001	3,830,472

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Wiking BW	Mile/Hrs:	126235			
Equipment Type	Bus	ID #:	GJ1Z3029		Baro:	28.17
Fuel Sp. Gravity:	0.824	Тетр:	104			
SG Corr Factor:	1.01				Time:	1550

RPM	Exh Temp	Pvalneh	(()	He	602	02	
2019	239	0.58	0	26	1.4	17.7	
2019	240	0.58	0	26	1.39	17.7	
2019	240	0.56	0	28	1.39	17.7	
2019	241	0.56	0	29	1.39	17.8	
2019	241.2	0.56	0	26	1.37	17.8	
2019	241.6	0.56	0	28	1.37	17.8	
2019.000	240.467	.567	.000	27.167	1.385	17.750	Mean
0	0.968848113	0.01032796	0	1.32916014	0.01224745	0.05477226	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.72E-05	0	0.01385	0.1775	28.9331757	438,511	3,940,530

3,979,935

**% Change PF= 3.90 %

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Hino (new)	Mile/Hrs	201840			
Equipment Type:	Bus	ID #:	2263		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2			
					Time:	

RPM	Exh Temp	Pv Inch	0.0	He	€02	O2	
2020	209.4	0.48	0	23.5	1.36	18.5	
2020	209.2	0.5	0	23.5	1.36	18.5	
2020	209.8	0.48	0	24	1.36	18.6	
2020	205.6	0.47	0	27	1.33	18.5	
2020	207.8	0.48	0	33	1.32	18.4	
2022	209.6	0.48	0	21	1.33	18.5	
2022	209	0.48	0	23.5	1.33	18.4	
			7				
2020.571	208.629	.481	.000	25.071	1.341	18.486	Mean
0.975900073	1.485164734	0.00899735	0	3.90969491	0.01772811	0.06900656	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
2.51E-05	0	0.01341429	0.184857143	28.9555113	453,369	4,318,399

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Hino (new)	Mile/Hrs:	262283			
Equipment Type	Bus	ID #:	2263		Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	104			
SG Corr Factor:	1.01				Time:	1510

RPM	Exh Temp	Pv Inch	CO	HC	CO2	02	
1978	234	0.44	0	29	1.32	17.9	
1978	238	0.44	0	28	1.32	17.9	
1978	239	0.44	0	28	1.3	17.9	
1978	242	0.44	0	28	1.3	17.9	
1978	242	0.44	0	31	1.3	17.9	
1978	241	0.44	0	28	1.3	17.9	
1978.000	239.333	.440	.000	28.667	1.307	17.900	Mean
0	3.076794869	0	0	1.21106014	0.01032796	0	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.87E-05	0	0.01306667	0.179	28.9267293	464,013	4,728,142

4,775,423

**% Change PF=

10.58 %

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Hino	Mile/Hrs	439958			
Equipment Type:	Bus	ID #:	GJ1Z657		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2		· <u>·····</u>	
					Time:	1015

RPM	Exh Temp	Pv Inch	CO	HC	CO2	O2	
2018	222	0.62	0	31.5	1.31	18.2	
2019	223	0.62	0	30	1.31	18.2	
2019	224	0.62	0	25.5	1.29	18.5	
2019	224.8	0.62	0	9	1.28	18.5	
2019	224.4	0.62	0	29.5	1.26	18.6	
2019	225	0.62	0	21.5	1.26	18.8	
2018.833	223.867	.620	.000	24.500	1.285	18.467	Mean
0.40824829	1.157007635	0	0	8.42021377	0.02258318	0.23380904	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
2.45E-05	0	0.01285	0.184666667	28.9456877	473,005	4,015,132

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Hino	Mile/Hrs:	503725			
Equipment Type	Bus	ID #:	GJ1Z657		Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	100			
SG Corr Factor:	1.01				Time:	1635

RPM	Exh Temp	Pv Inch	CO	HC	002	0)2	
2034	239	0.62	0	29	1.27	18.1	
2034	245	0.62	0	31	1.26	18.1	
2034	245	0.62	0	28	1.25	18.1	
2034	246	0.62	0	29.5	1.25	18	L.
2034	246.6	0.6	0	31	1.25	18	
2034	247	0.6	0	31	1.24	18	
							-
2034.000	244.767	.613	.000	29.917	1.253	18.050	Mean
0	2.940521496	0.01032796	0	1.28127541	0.01032796	0.05477226	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.99E-05	0	0.01253333	0.1805	28.9242685	483,128	4,185,823

4,227,681

**% Change PF=

5.29 %

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Hino	Mile/Hrs				
Equipment Type:	Bus	ID #:	GJ1Z1805		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2		5 544	1145
					Time:	1145

RPM	Exh Temp	Pv Inch	CO	HC	CO2	O2	
2019	219.6	0.6	0	22	1.33	18.5	
2019	218.6	0.64	0	20	1.32	18.6	
2007	220.4	0.64	0	25	1.28	18.4	
2007	220	0.62	0	25	1.27	18.4	
2008	220	0.64	0	22	1.25	18.5	
2008	220	0.64	0	22.5	1.26	18.5	
	×						
2011.333	219.767	.630	.000	22.750	1.285	18.483	Mean
5.955389716	0.625033332	0.0167332	0	1.94293592	0.03271085	0.07527727	Std De

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
2.28E-05	0	0.01285	0.184833333	28.9462528	473,422	3,974,684

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Hino	Mile/Hrs:	357460			
Equipment Type	Bus	ID #:	GJ1Z1805		Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	100			
SG Corr Factor:	1.01				Time:	1415

RPM	Exh Temp	Pv Inch	CO	HC	CO2	02	
2010	221	0.62	0	26	1.28	17.9	
2010	224	0.62	0	24	1.26	17.8	
2010	225	0.64	0	23	1.25	17.8	
2010	228	0.64	0	26	1.24	18	
2010	228	0.62	0	26	1.24	18	
2010	229	0.62	0	24	1.24	18	
2010.000	225.833	.627	.000	24.833	1.252	17.917	Mean
0	3.060501048	0.01032796	0	1.32916014	0.01602082	0.09831921	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.48E-05	0	0.01251667	0.179166667	28.9183737	484,906	4,100,083

4,141,084

**% Change PF=

4.19 %

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Hino	Mile/Hrs				
Equipment Type:	Bus	ID #:	GJ1Z1804		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:				

RPM	Exh Temp	Pv Inch	CO	HC	€02	02	
2024	222.2	0.54	0	24.5	1.43	18	
2024	222.6	0.54	0	22.5	1.41	18.1	
2024	223	0.52	0	55	1.38	18.3	
2024	224.6	0.54	0	28	1.37	18.3	
2024	225.2	0.5	0	58	1.36	19.1	
2024	224.6	0.52	0	49	1.33	18.5	
2029	224.2	0.52	0	17.5	1.31	18.5	
2024.714	223.771	.526	.000	36.357	1.370	18.400	Mean
1.889822365	1.157172251	0.01511858	0	16.9992997	0.04203173	0.36055513	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
3.64E-05	0	0.0137	0.184	28.9573087	441,746	4,071,909

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Hino	Mile/Hrs:	351629			
Equipment Type	Bus	ID #:	GJ1Z1804		Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	99.6			
SG Corr Factor:	1.01				Time:	1215

RPM	Exh Temp	Pv Inch	CO	HC	002	02	
2019	237.8	0.54	0	27	1.35	18.4	
2019	238.4	0.52	0	27	1.35	18.1	
2019	239	0.52	0	27	1.35	18.2	
2019	239	0.52	0	27	1.34	18.1	
2019	238.6	0.52	0	31	1.34	18.2	
2019	238.6	0.5	0	27	1.35	18.2	
2019	239	0.5	0	27	1.35	18.3	
2019	239	0.5	0	27	1.35	18.3	
2019	239	0.5	0	30	1.35	18.3	
2019	238.6	0.5	0	27	1.36	18.3	
2019.000	238.700	.512	.000	27.700	1.349	18.240	Mean
0	0.391578004	0.01398412	0	1.49443412	0.00567646	0.09660918	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.77E-05	0	0.01349	0.1824	28.9470466	450,167	4,250,388

4,292,892

**% Change PF=

Time:

1215

Company Name:	Gujarat	Location	Ambaji		Date:	11-Feb-94
Test Portion:	Baseline	Stack Diam.	2	Inches		
Engine Type:	Hino	Mile/Hrs	301670			
Equipment Type:	Bus	ID #:	GJ1Z1955		Baro	28.17
Fuel Sp. Gravity(SG	0.8330	Temp:	80.2		Time:	
					1 ime:	

RPM	Exh Temp	Pv Inch	CO.	HC	002	02	
2028	194.4	0.52	0	22.5	1.3	19.9	
2002	201.2	0.52	0	3.5	1.29	18.1	
2002	201	0.58	0	22	1.29	18.5	
2000	203.2	0.54	0	29.5	1.29	18.6	
2000	206	0.56	0	32	1.32	18.6	
2000	210	0.56	0	35	1.32	18.6	
2005.333	202.633	.547	.000	24.083	1.302	18.717	Mean
11.14749598	5.261812109	0.0242212	0	11.3287981	0.0147196	0.6112828	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw1	pf1	PF1
2.41E-05	0	0.01301667	0.187166667	28.9583302	467,319	4,158,468

Company Name:	Gujarat	Location:	Ambaji		Test Date:	5/31/94
Test Portion:	Treated	Stack Diam:	2	Inches		
Engine Type:	Hino	Mile/Hrs:	360727			
Equipment Type	Bus	ID #:	GJ1Z1955		Baro:	28.17
Fuel Sp. Gravity:	0.824	Temp:	100			
SG Corr Factor:	1.01				Time:	1700

RPM	Exh Temp	Pv Inch	CO	He	CO2	02	
2015	210	0.5	0	23	1.32	18.1	
2015	213	0.5	0	22	1.34	18.1	
2015	216	0.5	0	28.5	1.33	18	
2015	216	0.5	0	26.5	1.32	18	
2015	218	0.5	0	25	1.33	18.1	
2015	219	0.5	0	26.5	1.33	18.1	
2015	221	0.52	0	27.5	1.325	18.1	
2015.000	216.143	.503	.000	25.571	1.328	18.071	Mean
0	3.716116765	0.00755929	0	2.37045304	0.00698638	0.048795	Std Dev

VFHC	VFCO	VFCO2	VFO2	Mtw2	pf2	PF2
2.56E-05	0	0.01327857	0.180714286	28.9367974	457,542	4,288,176

4,331,057

**% Change PF= 4.15 %

Figure 2.

SAMPLE CALCULATION FOR THE CARBON MASS BALANCE

BASELINE:

Equation 1 (Volume Fractions)

VFHC = 13.20/1,000,000 = 0.0000132

VFCO = 0.017/100 = 0.00017

 $VFCO_2 = 1.937/100$ = 0.01937

 $VFO_2 = 17.10/100$ = 0.171

Equation 2 (Molecular Weight)

Mwt1 = (0.0000132)(86) + (0.00017)(28) + (0.01937)(44) + (0.171)(32)

+[(1-0.0000132-0.00017-0.01937-0.171)(28)]

Mwt1 =28.995

Equation 3 (Calculated Performance Factor)

pf1 =
$$\frac{3099.6 \times 28.995}{86(0.0000132) + 13.89(0.00017) + 13.89(0.01937)}$$

pf1 = 329,809

Equation 4 (CFM Calculations)

CFM =
$$\frac{(d/2)^2 \pi}{144} \left(1096.2 \sqrt{\frac{Pv}{1.325(Pb/Te+460)}} \right)$$

d =Exhaust stack diameter in inches

Pv = Velocity pressure in inches of H_20

Pb =Barometric pressure in inches of mercury

Te = Exhaust temperature ${}^{o}F$

CFM =
$$\frac{(10/2)^2 \pi}{144} \left(1096.2 \sqrt{\frac{.80}{1.325(30.00/313.100+460)}} \right)$$

CFM = 2358.37

Equation 5 (Corrected Performance Factor)

PF1 = 329,809(313.1 deg F + 460)

2358.37 CFM

PF1 = 108,115

TREATED:

Equation 1 (Volume Fractions)

VFHC = 14.6/1,000,000

= 0.0000146

VFCO = .013/100

= 0.00013

 $VFCO_2 = 1.826/100$

= 0.01826

 $VFO_2 = 17.17/100$

= 0.1717

Equation 2 (Molecular Weight)

Mwt2 =
$$(0.0000146)(86) + (0.00013)(28) + (0.01826)(44) + (0.1717)(32) + [(1-0.0000146-0.00013-0.01826-0.1717)(28)]$$

Mwt2
$$= 28.980$$

Equation 3 (Calculated Performance Factor)

pf2 =
$$\frac{3099.6 \times 28.980}{86(0.0000146) + 13.89(0.00013) + 13.89(0.01826)}$$

$$pf2 = 349,927$$

Equation 4 (CFM Calculations)

CFM =
$$\frac{(d/2)^2 \pi}{144} \left(1096.2 \sqrt{\frac{Pv}{1.325(Pb/Te+460)}} \right)$$

d =Exhaust stack diameter in inches

Pv = Velocity pressure in inches of H_20

Pb =Barometric pressure in inches of mercury

Te = Exhaust temperature ${}^{O}F$

CFM =
$$\frac{(10/2)^2 \pi}{144} \left(1096.2 \sqrt{\frac{.775}{1.325(29.86/309.02+460)}} \right)$$

$$CFM = 2320.51$$

Equation 5 (Corrected Performance Factor)

PF2 =
$$349,927(309.02 \text{ deg F} + 460)$$

2320.51 CFM

= 115,966

Fuel Specific Gravity Correction Factor

Baseline Fuel Specific Gravity - Treated Fuel Specific Gravity/Baseline Fuel Specific Gravity +1

$$.840 - .837 / .840 + 1 = 1.0036$$

$$PF2 = 115,966 \times 1.0036$$

$$PF2 = 116,384$$

Equation 6 (Percent Change in Engine Performance Factor:)

% Change PF =
$$\frac{PF2 - PF1}{PF1}$$
 x 100

% Change PF =
$$[(116,384 - 108,115)/108,115](100)$$

= $+7.65$

Note: A positive change in PF equates to a reduction in fuel consumption.

	Cur bon 1	1 100
Company: Cheran	Location: CBE Test Date: OS	94
Test Portion: Baseline	: Treated: Exhaust Stack Diameter:	Inches
Engine Make/Model: _ Type of Equipment: _	Leyland to 2 Miles/Hours: 169.208 TCB	<u> 7</u> 67 .
Fuel Specfic Gravity:	<u>1117</u> 0.832 @:_	89·2 (°F)

_____ inches of Mercury

Barometric Pressure:

RPM .	Exhaust Temp °E	P Inches	% CO	HC ppm	% CO ₂	% O ₂	NO.
2070.	258	0.5	0.02	15	1.63	17.5	
	258		0.02	15	1.64.	17.5.	
	259.		0.02	15	1.62	17.5.	- 64.5
	258		0.02	15	1.62	17.5	
	258.			15	1.62	17.5	
4	258			15	1.62	17.5	
	258.			15	1.62	17.5	1
	258.		0.02	19.	1.64	17.5	-
	258		0.02	lg	1.64	17.4	
2050	256.	0.5	0.02	19.	1.64.	17.4	

Names of Customer Personnel Participating in Test:

Fin 11.50

14. S. SR 1DHAQ. 84

102 14.2 1.629 12. ×8 (6. 754)

Signature of Technicians:

Start Time: 11. 45

Company: Cheran	Location: CBE		05/26/94	005
Test Portion: Baseline:		-	Diameter:Inches	and the second
Engine Make/Model: // Type of Equipment:	heyland 402 Miles/	163141 Hours:I.	D.#:_(498	82.6.
Fuel Specfic Gravity:	0.833		01(1	F)
Barometric Pressure:	inches of M	lercury	Start Time: 0	200.

RPM .	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO.
2025	234	0.44.	0.015	34	1.55	17.5	
	229		0.015	31	1.56	17.4.	
	230		0-015	31	1.57	17.4	
	230		0.015	37	1.57	17.4.	
	227	·	0.015	36	1.56	17.4	
	227		0.015	34	1.55	17.4.	
	227		0015	34	1.55	17.4.	
	230		0057	187	1.55	17.5	
	228		0.053	193	1.54	174	
4	230	0.42	0.05	20	1.54	174	

Smole. 1/6.0 2/5.0

Names of Customer Personnel Participating in Test:

FINISH. 0210-

10

19-8. SRID WROON

Z22 20 .43 Signature of Technicians:

(, 4838)



Company: CHERA	N Location: ComBAPORE Test Date: V 5.02.94	
Test Portion: Baseline:	Treated: Exhaust Stack Diameter:Inches	
Engine Make/Model: Type of Equipment:	HIND lemn: 1,81,555 Miles/Hours: I.D.#: JCB 1850	
Barometric Pressure: _	inches of Mercury (a): (°F)	5 moles
Fuel Specfic Gravity: _		2-9

Start Time: 2.20 AM

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2053	229.0	0.42	0.04	39	1.46	18-2	; }
	228.6	0.42	0.04	38	1.46	18-2	
	229	0.42	0.94	39	146	18.2.	
2049	228	0.42	0.04	39	1.42	18.3	
	228	0.42	0.04	40	1.42	18.3	
	228.2	0.42	0.04	40	142	18.3.	

2065	229.8	0.42	0.04	40	1-41	184	2 1
	229.8	0.42	0.04	42	1-42	Finish Time:_	2.4 5 mg
	229.8 Signa	0.42 ture of Tecl	o o4	40	1.42	18.4.	

104 37.89 1.432



Company: CHERAN Test Portion: Baseline:	Location: Location: Location: Location: Location: Exhaust Stack Diameter:Inches	ès
	370 Miles/Hours: 1,56,81 P.D.#: 1443	
Barometric Pressure:	inches of Mercury @:(°F)	Smoles 1-9.5
Fuel Specfic Gravity:	0.826 @: <u>88.2</u> (°F)	2-6.5
	Start Time:	1.25 Am

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2040	2284	0.44	0.09	40	1.71	17.8	\
	228.4	0.44	0.09	40	1.71	17.8	
	2282	0.44	0.09	40	1.70	17-8.	
2040	228.0	0.44	0.09	44	1.68	17.9	
	228.4	0.49	0.09	43	1.67	17.9	
	228-4	0.44	0.09	44	1.67	17.9	

2041	228-2	0.42 0.42 ture of Tech	0.09	43	1.66	17.8	1.55Am
	Signat	ture of Tech	nicians:		1.37	11.8	

109 42.22 1.682



Water Press ____ Oil Press

Start Time: 11.50 Pm

LHL		Del Company
SGA	Company: Cherun Location Combactore Test Date: Feb. 4 94 Test Portion: Baseline: X Treated: Exhaust Stack Diameter: Inches	
	Test Portion: Baseline: Treated: Exhaust Stack Diameter:Inches	2 . C L
	Engine Make/Model: 37.0 Miles/Hours: 153,446 I.D.#: 1584 Type of Equipment:	Smoke Spot 1- 4.5 2-6.5
	Barometric Pressure: inches of Mercury @: (°F)	
	Fuel Specific Gravity: 824 @: 928 (°F)	

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
1977	219.2	0.10	0.07	49	1.64	17.9	
* 3	219.6	0.12	0.07	49	1.62	17.9	
	219.8	0.12	0.07	49	1.62	17.9	
1993	219.6	0.14	0.07	52	1.61	180	
	219.8	0.14	0.07	52	1.59	18.)	
	219.6	0.14	0.07	52	1.59	18.)	

223.0 0.14 0.07 1988 222.6 0.14 0.07 18.0 Signature of Technicians: 1.57 18-0 221.6 52 1995 0.07 1.58 18.1 221.6 53 0.07 18.1 18.1

55.25 1.592 180 Names of Customer Personnel Participating in Test:

Company: CHERAN	Location: COIMBATORE Test Date: 05.02.94.	
Test Portion: Baseline:	Treated: Exhaust Stack Diameter:Inches	
	402 Miles/Hours: I.D.#: TCB 1498.	
	inches of Maraury (9): (9E)	Smore 1-2:5
Fuel Specfic Gravity:0	@: <u>827.</u> (°F)	2-2.

Start Time: 4.25 AM P Inches % CO₂ % O₂ NO_x **RPM** Exhaust % CO HC Temp °F of H₂O ppm 0.46 18.2 229 0.05 38 1.51 2027 229 182 0.05 38 1.51 0.46 229 0.05 18.2 38 1.51 0.46 18.2 1.51 0.06 0.46 35 2035. 229.2 1.51 0.06 35 18-2 0.46 229.2 0.46 0.05 18-2 229.4 35 149

222. 30 .46 .053 36.7 1.507 Finish Time: 4.40 AM

Signature of Technicians:



Company: CHERAN I	Location: OS IMBATORE Test Date: 15-02 94.	
Test Portion: Baseline:		
Engine Make/Model:	402 Kms: 1,39,865 Miles/Hours: I.D.#: TC13 1767	
Barometric Pressure:	inches of Mercury @:(°F)	Smorce
Fuel Specfic Gravity:	(a): X 21. CH	1-92-6

Start Time: 3110 AM

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
20.74	238	0.54	0.03	27	1.67	18-2	٢
	238-4	0.54	0.03	27	1.67	18-2	
r Ca	238.6	0.54	0.03	27	1-67	18.2	
2076	240	0.52	0.03	28	1-63	18.0	
	240.2	0.52	0.03	28	1.63	18.0	
	240	0.52	0.03	3.0	1.63	120	<i>J.</i>

Finish Time: 3.25AM

Signature of Technicians:

Names of Customer Personnel Participating in Test:

1

Company: WER	Location: U	B€ Test Date:_	105.02.94	
Test Portion: Baseline		d: Exhaust Stack		
Engine Make/Model: ₋ Type of Equipment: _		Miles/Hours:I	B.#: FCB 1504	
Barometric Pressure:	incl	nes of Mercury @:	(°F)	Imolee 1-6
Fuel Specfic Gravity:	0.828.	@: <u>\</u>	2-2- (°F)	2-6

Start Time: 4.05Am

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2059	230	0.40	0.04	37	1.72	17-6	T
	230	0.40	0.00	37	1.72	17-6	
	229.8	0.40	0.04	28	1.73	17.6	
2058	231-8	0.40	0.04	40	1.69	1778	
	232	6.40	0.04	39	1.69	17-9	
	232	0.40	0.04	40	1.69	17.9.	+

Finish Time: 4.25Am

Names	of Customer	Personnel Par	ticipating in Test:	



Company: Cheran Location: Coimbatore Test Date: 766.4,194 Test Portion: Baseline: X Treated: Exhaust Stack Diameter: Inches	
Engine Make/Model: 402 KM 254,655 I.D.#: 4921 Type of Equipment:	Snoke no.
Barometric Pressure: inches of Mercury @: (°F)	1. above 9
Fuel Specfic Gravity:	

Start Time: 12-35 Am

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2012	2364	0.44	0.04	32	1.5)	18.0	
	236.4		6.04	32	1.50	18.0	
	236.4	a .	6.04	32	1:50	18.0	
2015	236,8		0.04	34	1.50	18.2	
	237.2	0.42	0.04	34	1.48	18.3	
	239.8		0.04	33	1.48	183	

238.6 0.44 0.04 34 1.48 Finish Time: 12.50mm 237.4 0.04 34 1.48 18.3 Signature of Technicians: 237.6 0.42 0.04 33 1.48 18.3



Company: WERAN Locat	ion: COLMRATORE Test Date: 05.02.94	
Test Portion: Baseline:	Treated: Exhaust Stack Diameter:Inches	
Engine Make/Model: 370. Type of Equipment:	Miles/Hours:I.D.#:_1081544	
Barometric Pressure:	inches of Mercury @:(°F)	Smola 1-85
Fuel Specfic Gravity: 0.8	@: <u>85.8</u> (°F)	2-7.5.

Start Time: 1. 55 Am

RPM	Exhaust	P Inches	% CO	НС	% CO ₂	% O ₂	NO _x
KIN	Temp °F	of H ₂ O	70 00	ppm	70 CO2	70 02	NO _x
2032	240	0.44	0.06	55	1.67	17.7	
	239	0.44	0.06	55	1.67	17-7	
	240	0.44	9.96	55	1-67	17.7	
2035	238	0,44	0.06	52	1.63	17-9	
	238	0.44	0.06	52	1.63	17.9	
	237.8	0.44	0.06	52	1.63	17.9	ł
	0041				-		

2035	234-6	0142	0.06	53	1.61	Finish Time:	2:15AM
	233.8	0.42	0.06	53	1.91	18.5	/
	233.4 Signa	or42	0.06	53	1.61	18.0	

Company: LHERAN Location: CAMBATORE Test Date: 0502.94.	
Test Portion: Baseline: Treated: Exhaust Stack Diameter:Inches	
Engine Make/Model: 370 Miles/Hours: I.D.#: 1,74,952 Type of Equipment:	
Barometric Pressure: inches of Mercury @: (°F)	Smolce 1-4
Fuel Specfic Gravity: 0.827 @: 85.2 (°F)	25

Start Time: 2.50 Ary

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2022	234	0.34	0.06	43	1.63	180	1
	234.4	0.36	0.06	43	1-63	180	
	234.8	0-36	0.06	43	1.67	18.0.	
2064	236.4	0.36	0.06	42	7.61	17.9	
	2.3.6.8	0.36	0.06	43	1.61	180	
	236.6	0.36	0.06	43	1.60	18-0	

Finish Time: 3.05Ay

Signature of Technicians:

Names of Customer Personnel Participating in Test:

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UHI SGA -9000 Carbon Mass Balance Field Data Form

Water Temp of Oil Press. to

Company: Cheran Locat	ion: Oimbatore Test Date: Feb 4, 44
Tost Dortion: Dosalina:	Treated: Exhaust Steels Diameters Inches
Engine Make/Model: HINO Type of Equipment: Bus	Kun Miles/Hours: 22928 I.D.#: 1889 Smoke Spot
Barometric Pressure:	inches of Mercury @: (°F)
Fuel Specfic Gravity: 1823	@:(°F)
0-1	Ambrent 82.3 Start Time: 11:05

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2014	197,8	15	,03	28	1.25	18.4	
1887	198.4	2.10	£03	28	1.24	18.5	
2.8	198.6	. Ole	·03	28	1.24	185	
2006	199.2	.08	103	27	1.25	18.6	les de la constante de la cons
Ta\	199.0	.08	.03	27	1.25	18.6	
	199.2	:08	:03	27	1.26	18.6)

Finish Time: 11:15

Signature of Technicians:

Company: Cheran Locati	Treated: Exhaust Stack Diameter:Inches	
Test Portion: Baseline:	Treated: Exhaust Stack Diameter:Inches	
Engine Make/Model: HINO Type of Equipment:	Miles/Hours:I.D.#:_0483	
Barometric Pressure:		Smaler Spot
Fuel Specfic Gravity:0.825	@:_88.~ (°F)	2-4.5-

Start Time: 1.05 PM

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
1988	214.8	0.62	0.05	32	1.54	18.0	ĺ
	214.8	0.62	0.05	32	1.54	18.0	
	214.6	0.62	0.05	32	1.55	17.9	
1988	216	0.64	0.04	32	1.54	18.1.	
	216.4	0.64	0.04	32	1.53	18.1	
	216.4	2.64	0.04	32	1,53	18.)	

Finish Time: 1.15 AM

Signature	of	Techi	nicians:
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Company:	CHERAN	Location: USE	Test Date	5. R.94	***************************************	
Test Portion:			Exhaust Stack	No.		
		HINO Mil	cm's: 2,58,9	168	- 0	
Engine Make/	Model:	It INO Mil	es/Hours:	J.D.#: <u>√√</u>	039.	
Type of Equip	ment:		/	<u>/</u>		
D	C	16 Million inches of	28.53	(OE)		Smole
Barometric Pr	essure:	inches of	Mercury $w:$	(°F)		1-5
Fuel Specfic (Gravity:	0.825	@:	86.0 (°F)		2-4

RPM	Exhaust Temp °F	P Inches of H ₂ O	% CO	HC ppm	% CO ₂	% O ₂	NO _x
2004	219.4	0.52	0.04	35	1.35	1893	
	219.8	0.52	0.04	34	1.35	18.4	
Sec.	220	0.52	0.04	35	1.34	184	
2015	221.6	0.54	0.04	32	1.34	185	
	222	0.54	0.04	32	1.33	18.5	
T T	222	0.54	0.04	32	1.34	185.	T T

Finish Time: 4.00

Signature of Technicians: